



Modular Infrastructure for Rapid Flight Software Development

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Overview

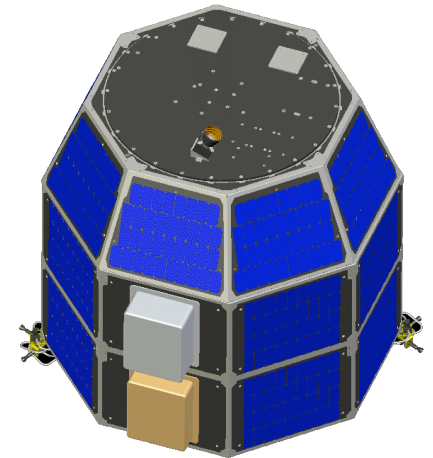


- Background
- Flight Software Development Process
- Simulink Model Overview
- Integration with cFE



Background

- Small Spacecraft Investigation
 - Modular CommonBus Spacecraft
- Hover Test Vehicle (HTV) Development
- Next Step - Lunar Atmosphere and Dust Environment Experiment (LADEE)
 - Joint ARC/GSFC Mission
 - Lunar Orbiter, Launch 2012





Flight Software Infrastructure Development

- Model Based Approach for Application Unique Software
- Latest Developments
 - Mathworks Simulink/RTW Embedded Coder
 - Integration of GSFC ITOS GDS Tool
 - Integration of GSFC Core Flight Executive (cFE)
 - Demonstrated on HTV



Hover Test

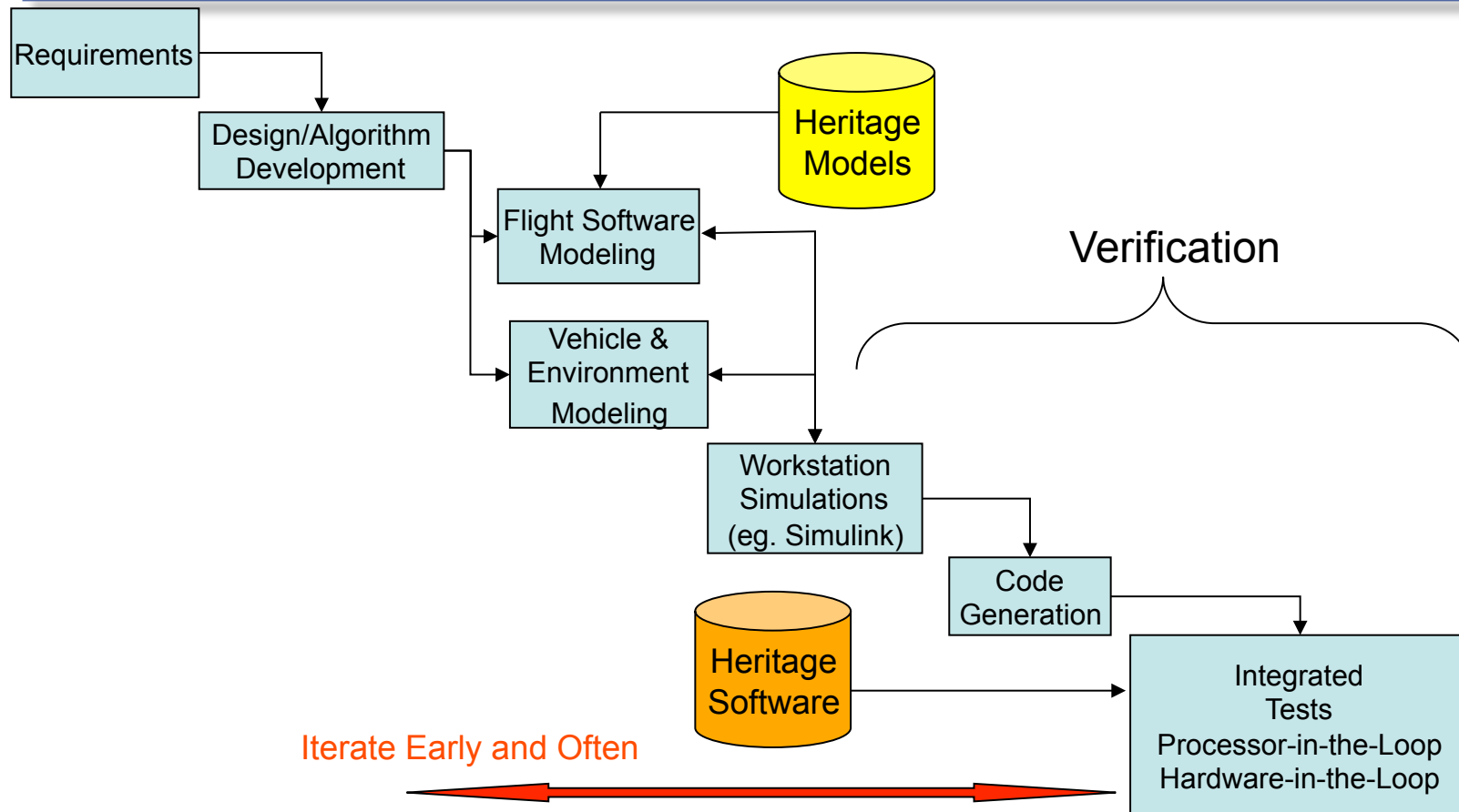




Flight Software Development Process Overview



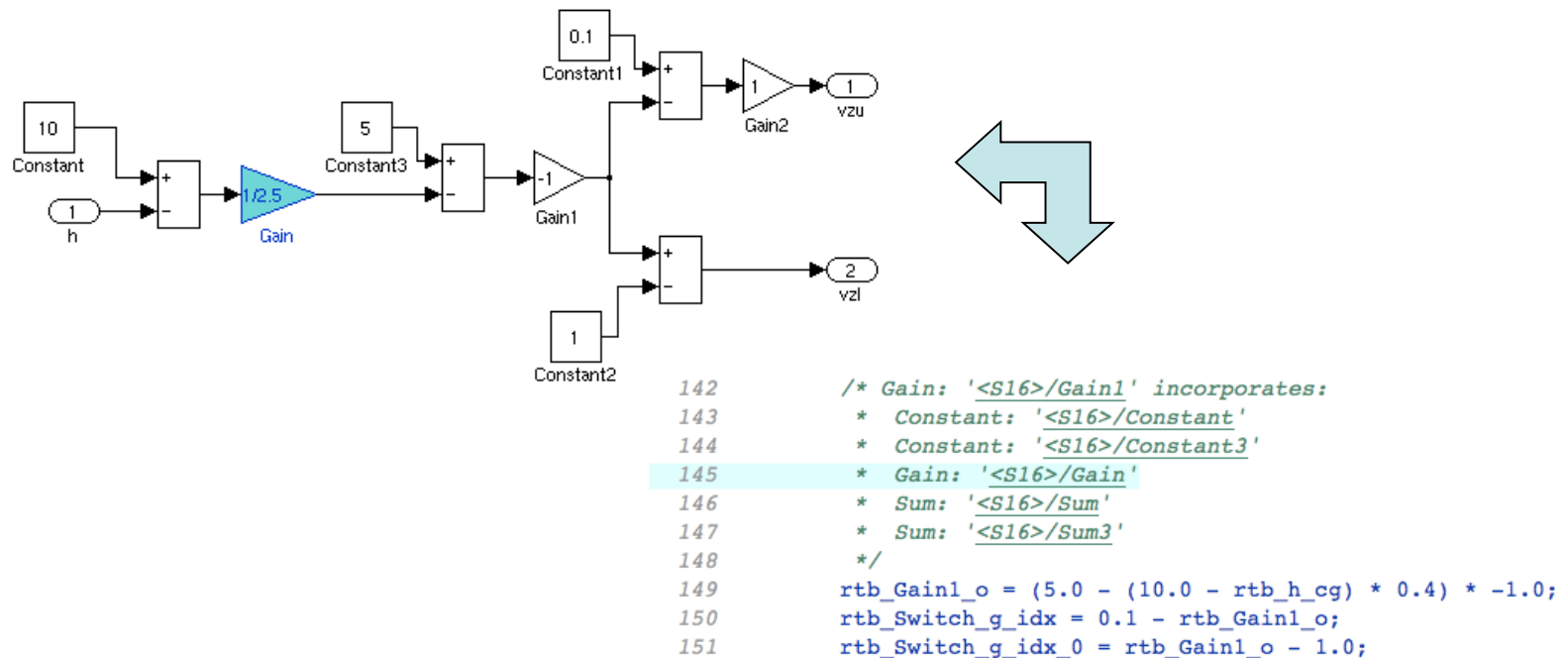
FSW Process Overview



- Model Based Development Approach
 - Develop Models of FSW, Vehicle, and Environment in Simulink
 - Automatically generate Software using RTW/EC.
 - Integrate with hand-written and heritage software.
 - Iterate while increasing fidelity of tests – Workstation Sim (WSIM), Processor-In-The-Loop (PIL), Hardware-in-the-Loop (HIL)



Automatic Code Generation



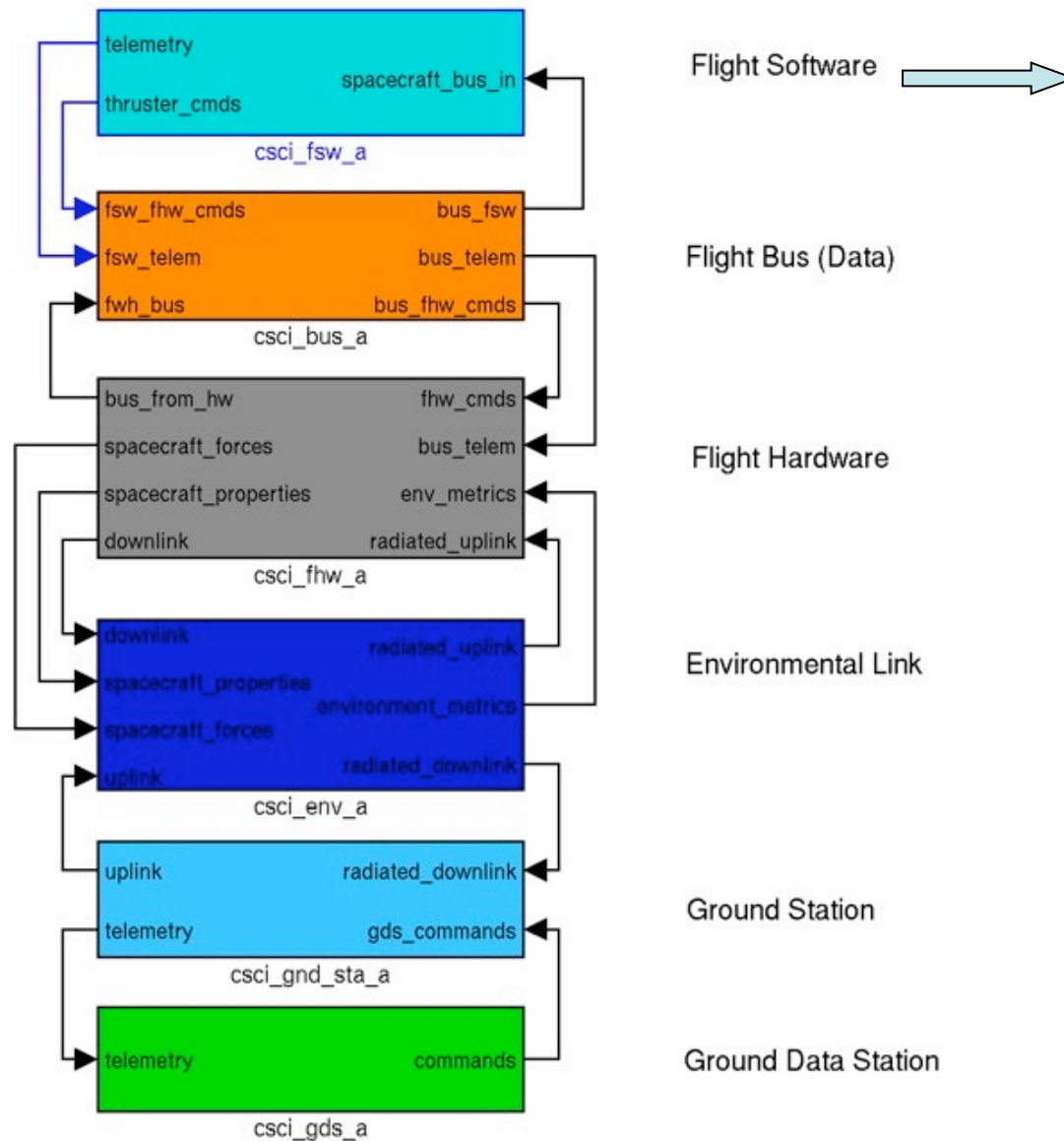
- Simulink supports two way trace-ability between models and generated code
- Code Easy to read, well commented



Simulink Model Overview



Simulink HTV Architecture



**FSW Auto-Coded
and integrated with
CFE**



Simulink FSW Model

Command Processing:

- Receives commands via CDH (TCP/IP or RS422).
- Compiled in script allows flexible sequencing.
- Processes and Sets Control Modes.

Vehicle Health Monitoring:

- Command Checking
- Sensor Limit Checking
- Hardware status

State Estimation:

- Receives sensor data.
- Low Pass Filters
- Auto generated Kalman Filter.

Telemetry:

Passes data to the CDH so that it can be transmitted via TCP/IP or RS422.

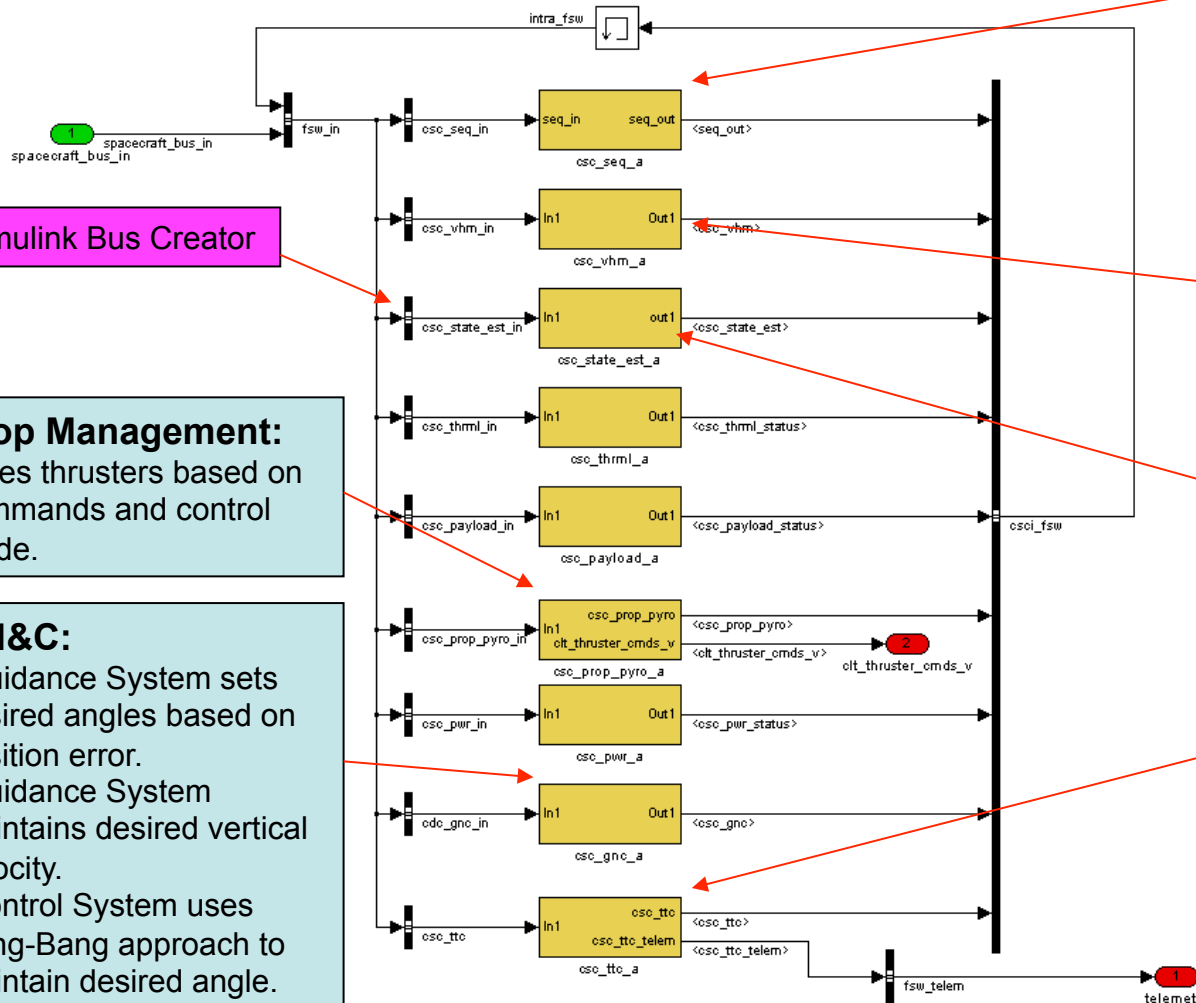
Simulink Bus Creator

Prop Management:

- Fires thrusters based on commands and control mode.

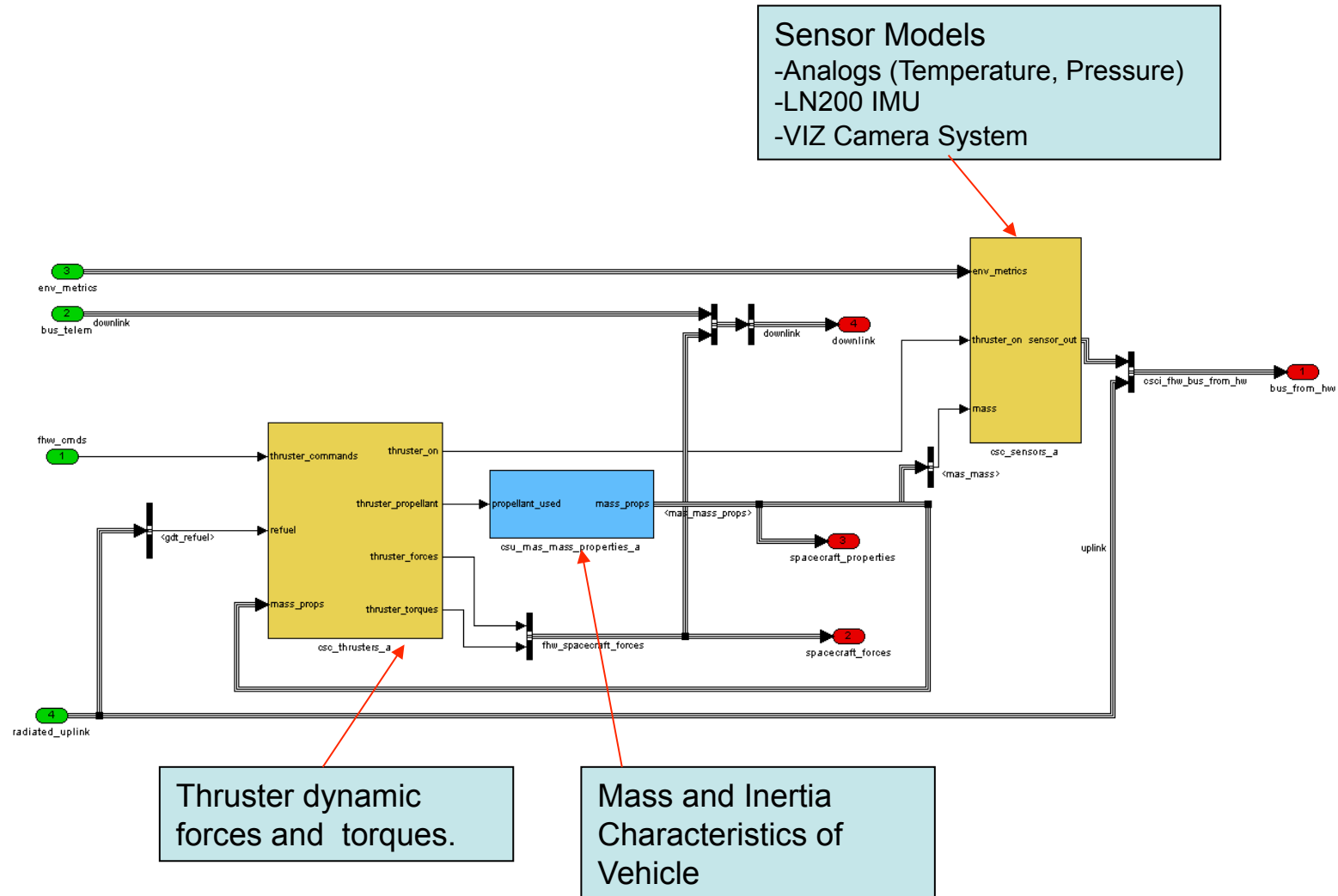
GN&C:

- Guidance System sets desired angles based on position error.
- Guidance System maintains desired vertical velocity.
- Control System uses Bang-Bang approach to maintain desired angle.



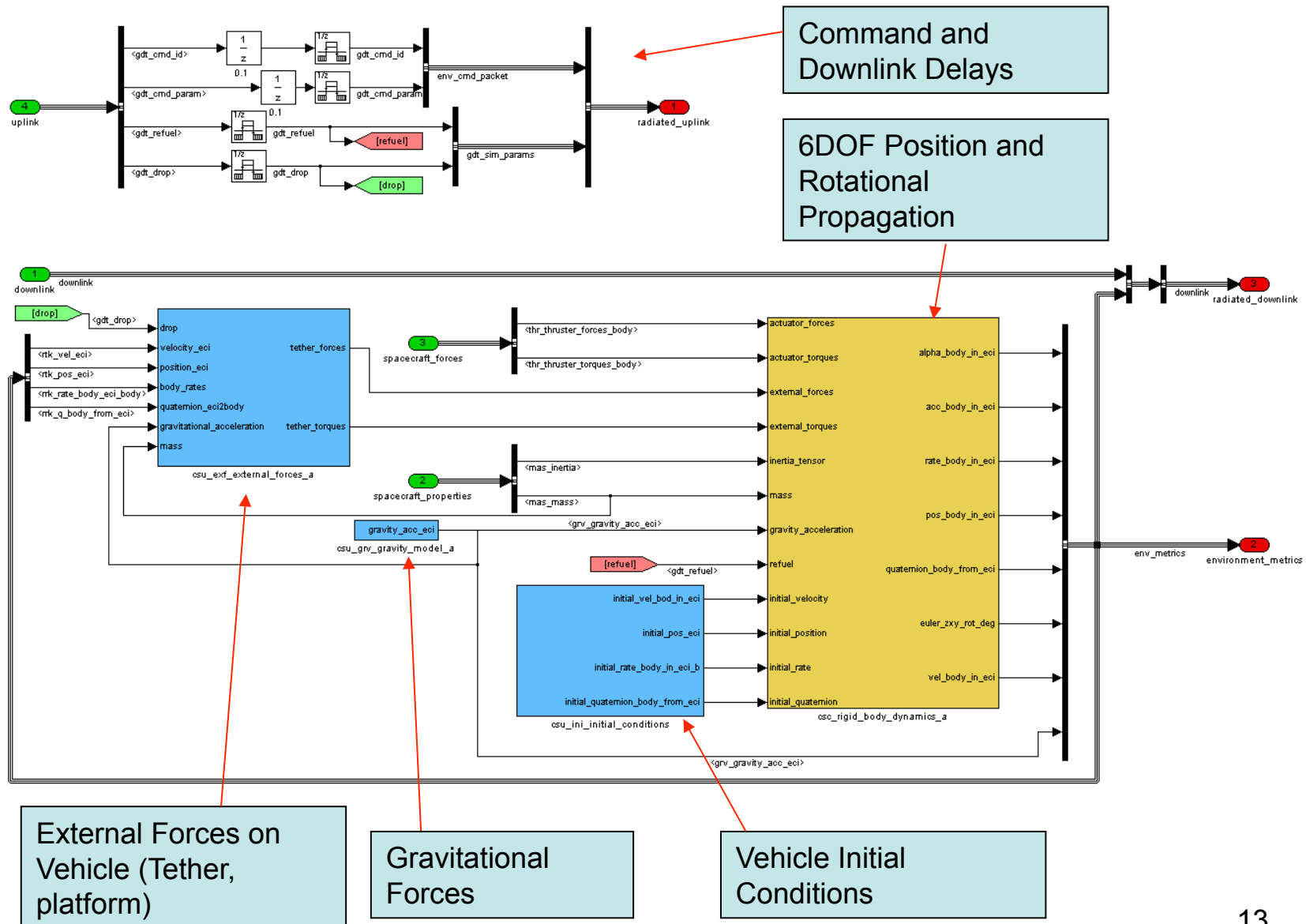


Simulink Flight Hardware Model





Simulink Environment Model





cFE Simulink Integration



cFE – Core Flight Executive

- Goddard Space Flight Center Developed
- Derived from Legacy Missions
- Flexible infrastructure for Space Flight Software
- Components:
 - Executive Services
 - Event Services
 - Time Services
 - Table Services
 - Software Bus Services

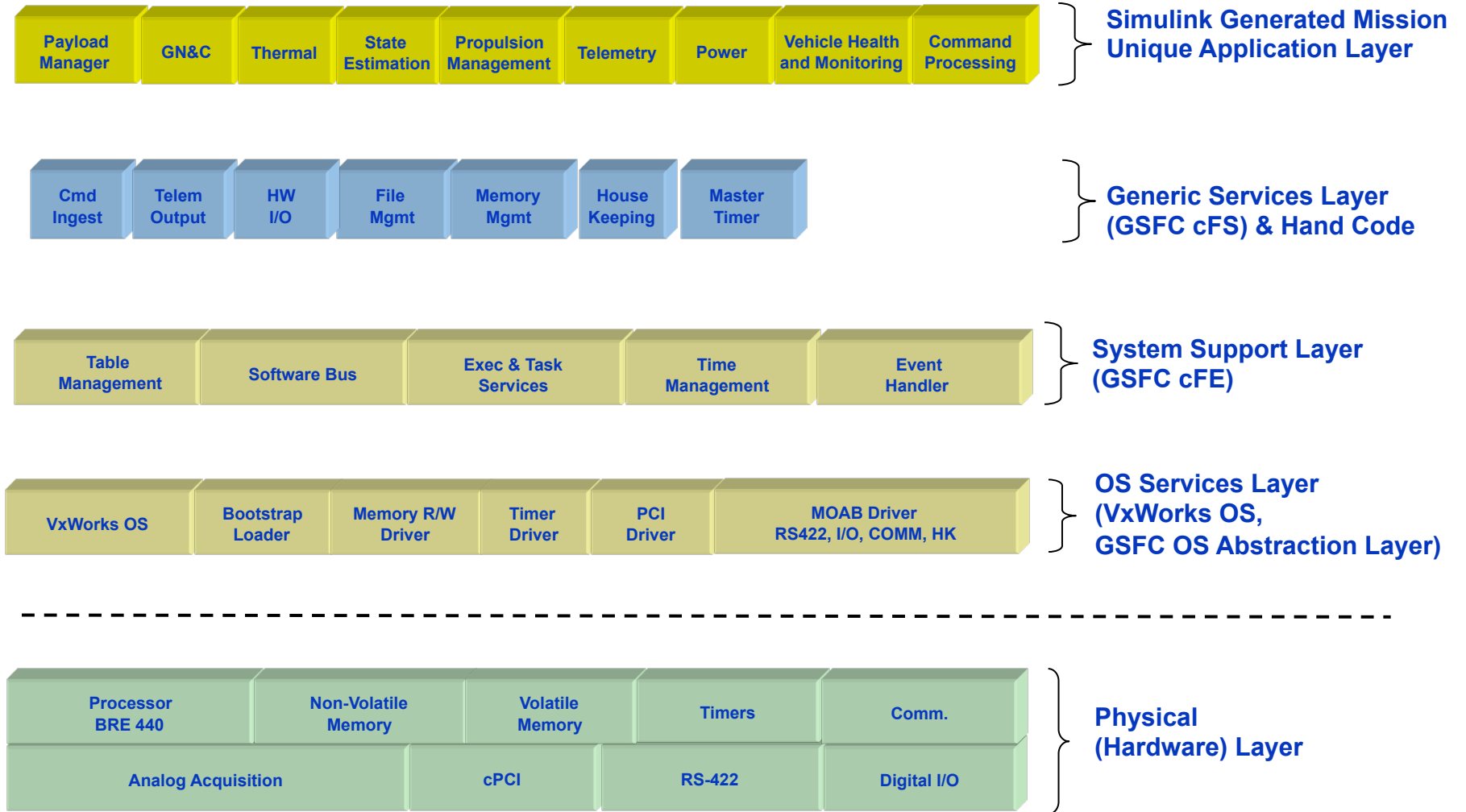


cFE Simulink Development Goals

- Utilize cFE with no changes
- Automate process during Code Generation.
- Subsystem Blocks generate to cFE Applications that run at desired rates
- Simulink Apps/Blocks Communicate via cFE Software Bus



Layered Architecture Approach





cFE Simulink Key Ideas

- Modular Tasks (vs. Monolithic)
 - Pros:
 - More Flexible
 - Simplifies Task Replacement
 - Easier Debugging – can look at messages between tasks
 - Cons:
 - Harder to implement
 - More overhead due to more tasks and messages
- Mathworks Template (TLC) File
 - Executed during Code Generation Process
 - Allows customization of created code
 - Leveraged to autocode cFE Apps from Simulink

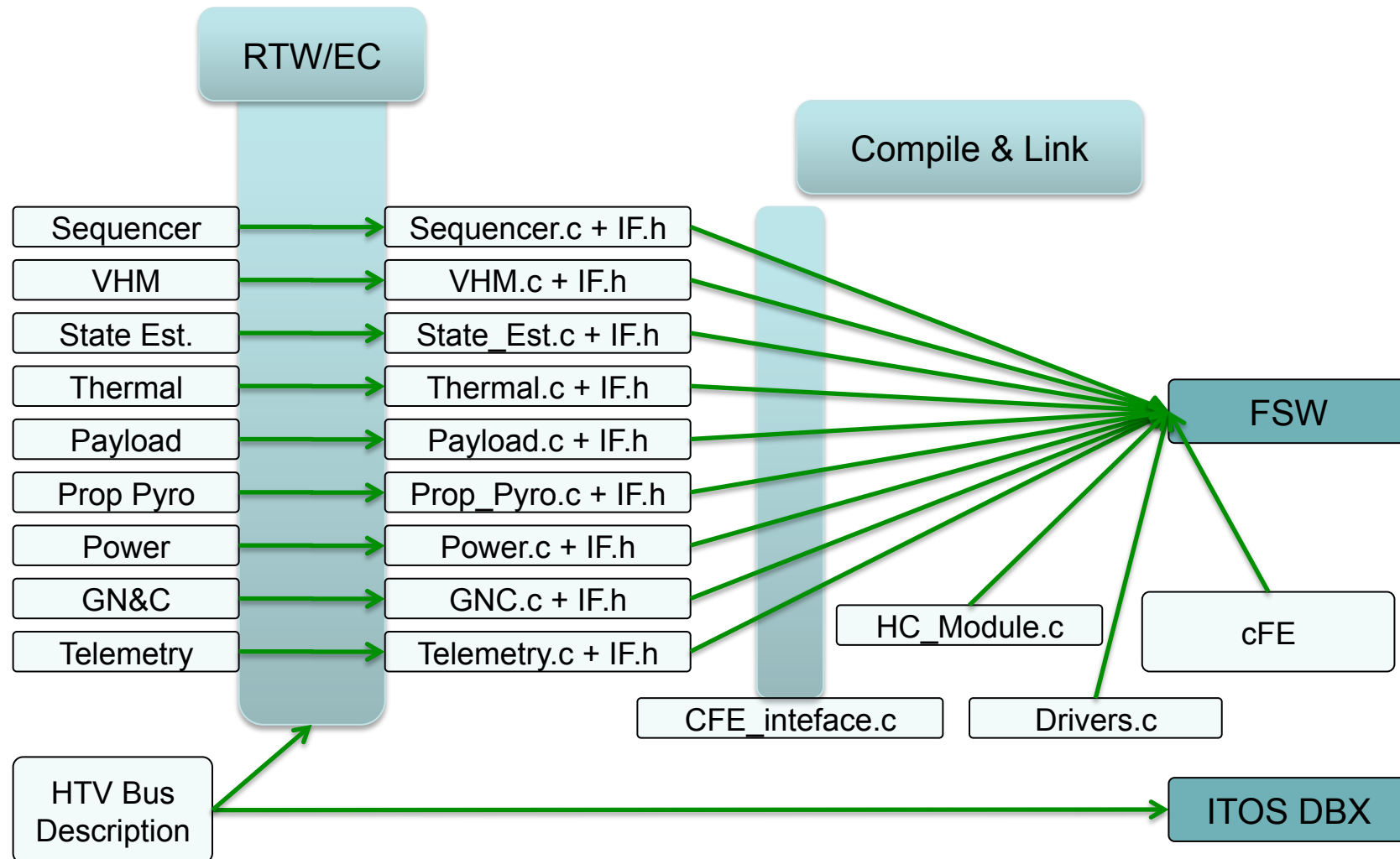


cFE Simulink Implementation

- Simulink Bus translates to cFE Message
- RTW/EC generates Task Description
- Master Timer Generates “Tick” to Schedule Apps and generate Output Messages
- Receive Structure Msgs update local App Input Values
- Apps also Respond to Other Command and Housekeeping Messages



cFE Simulink Autocode Process





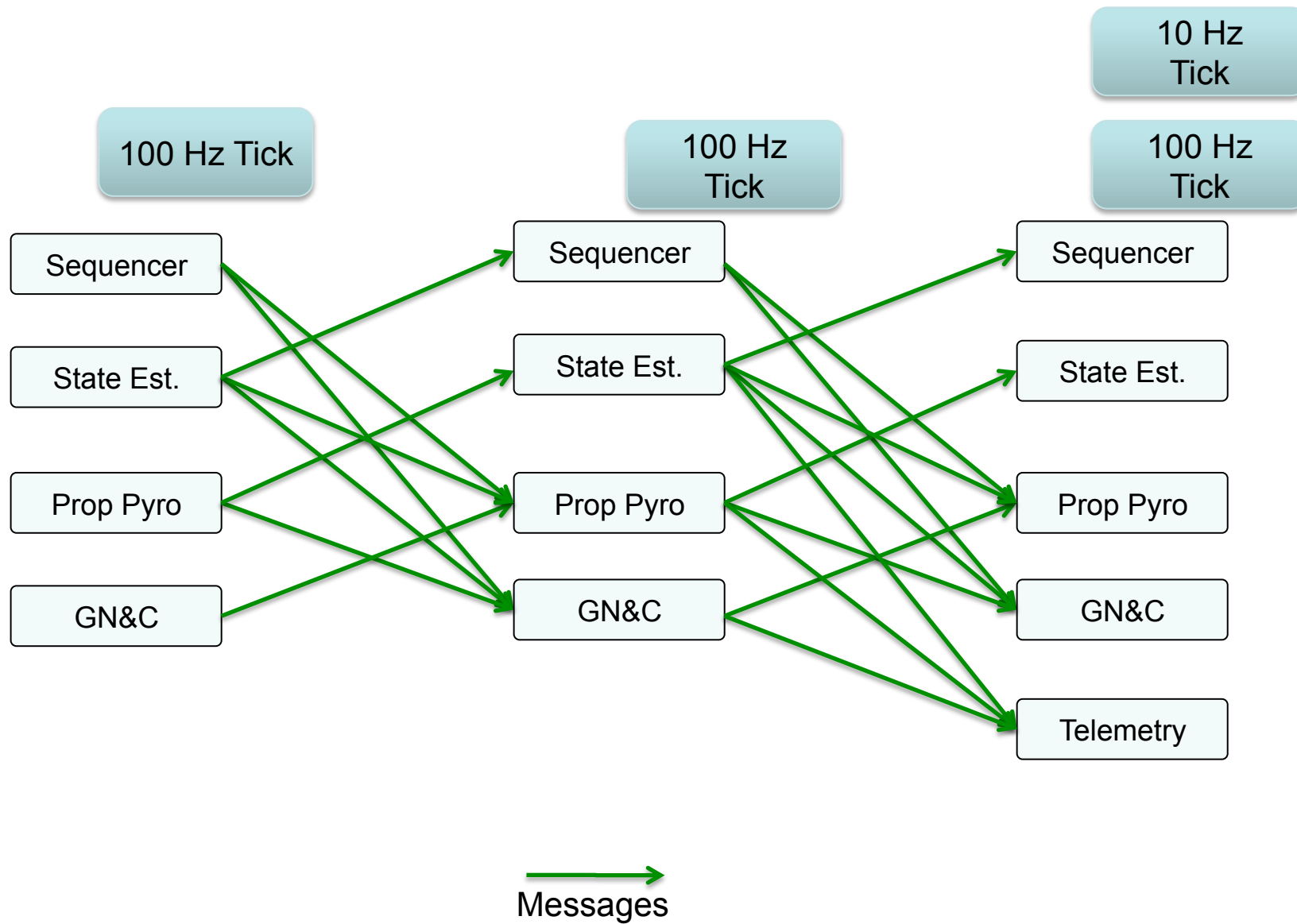
Simulink Bus becomes cFE Message

The screenshot shows the Simulink BusCreator and Properties windows. The BusCreator window on the left shows the 'Parameters' tab with 'Inherit bus signal names from input ports' selected and 'Number of inputs' set to 2. The 'Signals in bus' list contains 'csc_sensor_out' and 'uplink'. The 'Bus object' is 'csci_fhw_bus_from_hw'. A blue arrow points from the BusCreator window to the Properties window on the right. The Properties window shows the 'Simulink.BusElement: lns_delta_velocity_counts' configuration. The 'Name' is 'lns_delta_velocity_counts', 'DataType' is 'int16', 'Mode' is 'Built in', 'Complexity' is 'real', 'Dimensions' is 3, 'SamplingMode' is 'Sample based', and 'SampleTime' is -1.

```
'lns_msg', ...  
", ...  
sprintf(""), { ...  
    {'lns_delta_velocity_counts', 3, 'int16', -1, 'real', 'Sample'}; ...  
    {'lns_delta_angle_counts', 3, 'int16', -1, 'real', 'Sample'}; ...  
    {'lns_status', 1, 'int16', -1, 'real', 'Sample'}; ...  
    {'lns_mode', 1, 'int16', -1, 'real', 'Sample'}; ...  
    {'lns_data', 1, 'int16', -1, 'real', 'Sample'}; ...  
    {'lns_counts', 3, 'int16', -1, 'real', 'Sample'}; ...  
    {'lns_checksum', 1, 'int16', -1, 'real', 'Sample'}; ...  
} ...
```



cFE Simulink Message Flow





cFE Simulink App Loop

```
Struct App_Inputs In
Struct App_Outputs Out
App_Init() {
    Initialize_App_Inputs()
    Subscribe_SB_Msgs(Tick, AppMsgs,...)
    Simulink_Init(In, Out)
}
App_Main(){
    App_Init()
    while(1) {
        sb_receive_msg(msg, timeout)
        if (msg == tick) {
            Simulink_Step(dt, In, Out)
            sb_send_msg(Out) /* app update */
        } else {
            If (msg == app_update) /* Process other App Msgs */
                App_Update_Inputs(msg, Out)
            else Process_Msg(msg) /* HK, Cmds, etc... */
        }
    }
}
```



New Efforts

- 3DOF Simulator
- Command Queueing
- Parameter Tables
- Command & Telemetry Dictionary - XTCE
- Snapshot/Snapshot Recall
- Latency Reduction
 - Output Message triggers “Step” of Next Module
 - Retains Modularity



Summary

- NASA Ames developing infrastructure for rapid flight software development
- Model based process leverages Mathworks Simulink, RTW-EC
- Developed modular approach to integrate auto-generated code with GSFC's cFE.
- Successfully demonstrated on HTV
- Being Utilized on NASA's LADEE mission



Backup



cFE IMU App Loop

```
IMU_Main(){
    while(1) {
        struct imu_input_str imu_in
        read_msg_que(imu_in, timeout) /* VxWorks Msg Que */
        sb_send_msg(imu_msg)
        Send_tick()
    }
}

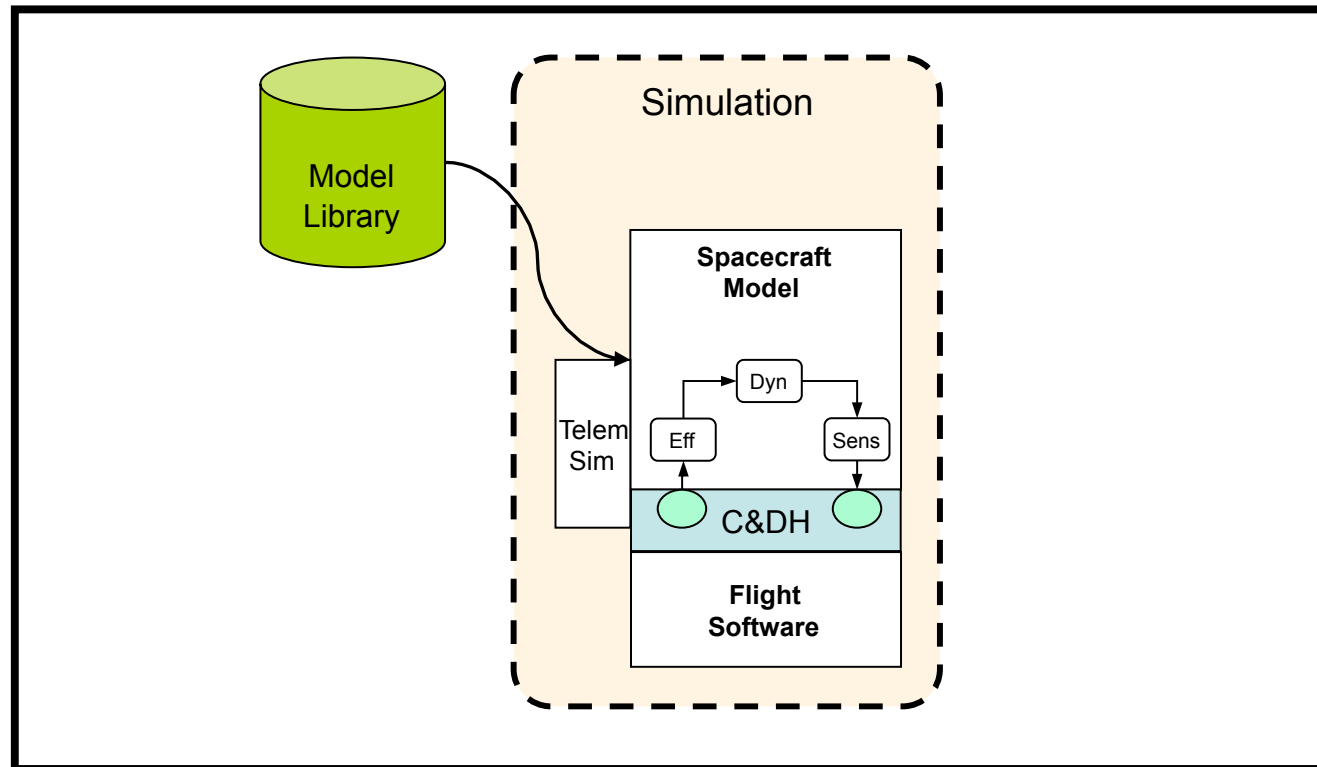
Cnt = 0;
Send_tick() {
    sb_send_msg(400HZ_Tick) /* Do we need 400HZ Tick or key off of IMU Data? */
    if ((Cnt % 2) == 0) sb_send_msg(200HZ_Tick)
    if ((Cnt % 4) == 0) sb_send_msg(100HZ_Tick)
    if ((Cnt % 40) == 0) sb_send_msg(10HZ_Tick)
    if ((Cnt % 400) == 0) sb_send_msg(1HZ_Tick)

    Cnt++;
}

/* Note: Other Apps same as IMU without the Send_tick() */
```



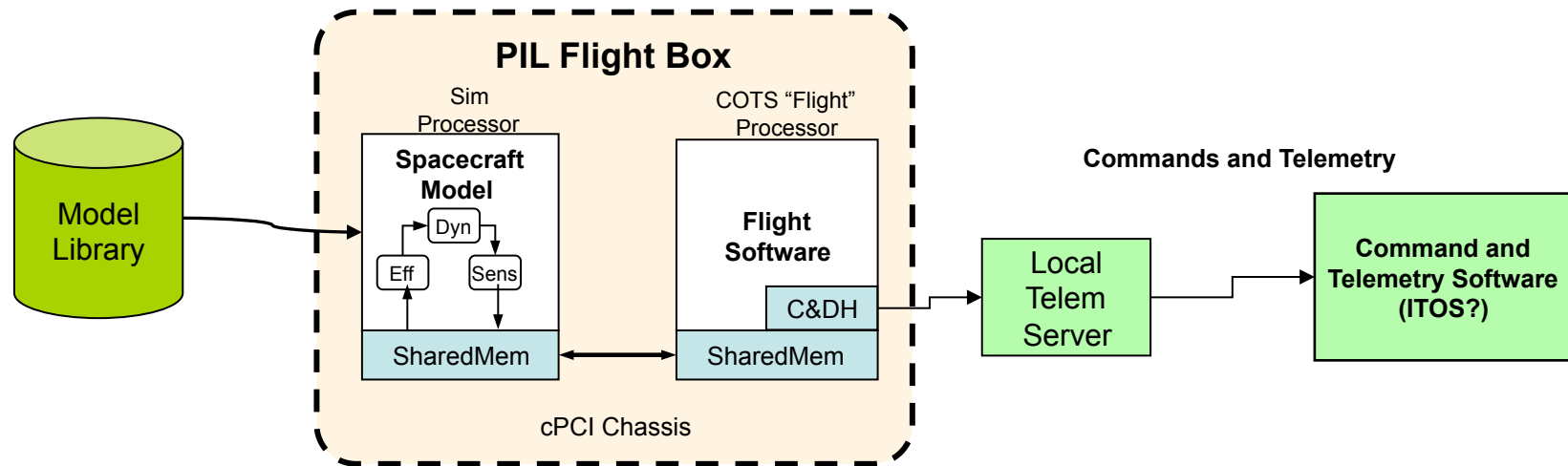
Workstation Simulation



- Simulink/SystemBuild Only (No Autocode)
- Early in development process
- Algorithm Development
- Requirements Analysis



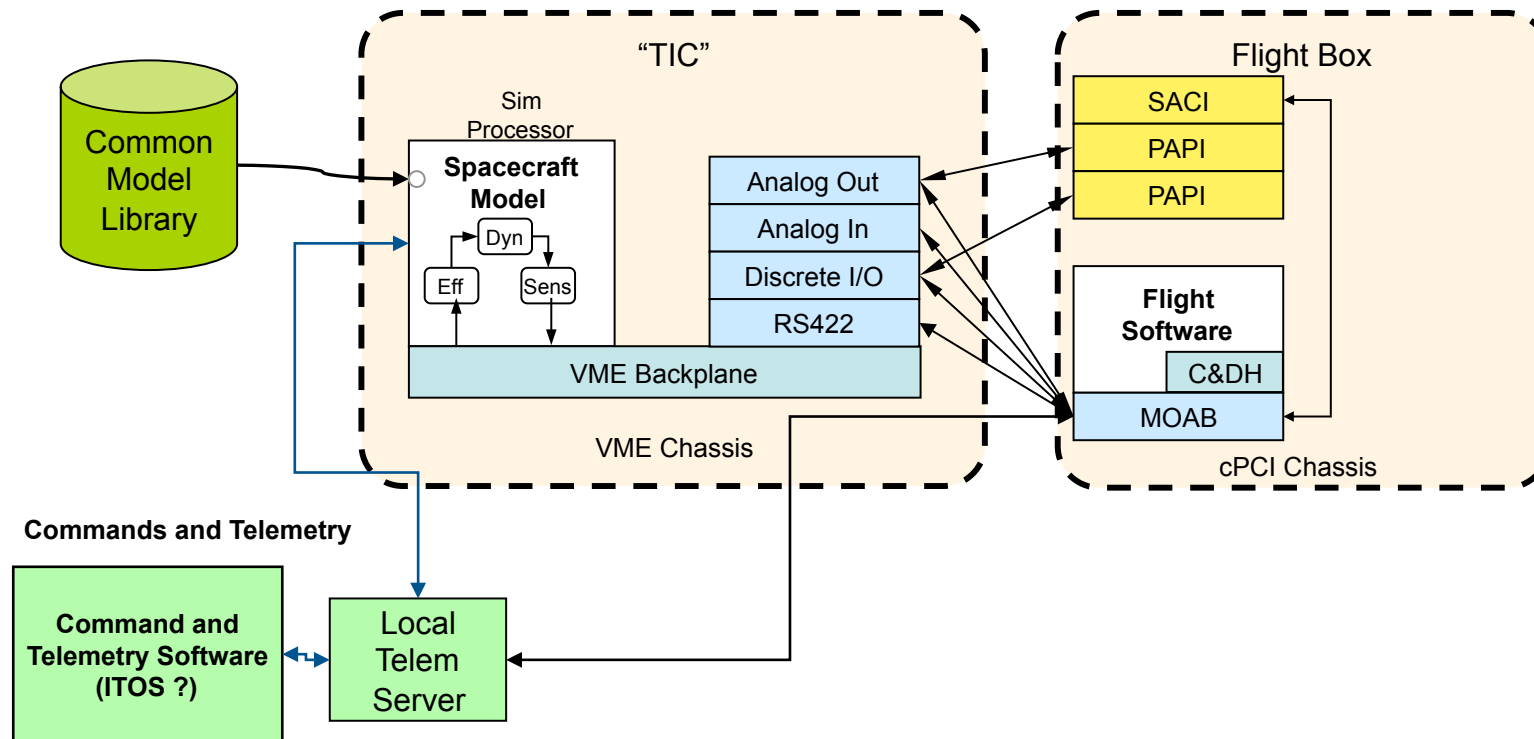
Processor-in-the-Loop Simulation



- Models autcoded and running on RT processors
- Inexpensive “flight-like” processor
- Tests autocoding process & integration with C&DH software
- Integration with Telemetry Software allows early development/testing of downlink
- Can be used for initial code size and resource utilization analysis



Hardware-in-the-Loop Simulation



- Flight code runs on Flight Avionics EDU
- Provides testing of FSW with Avionics I/O
- Definitive answers on resource utilization
- Highest fidelity simulations for verification/validation



Motivation for Moving to Simulink

- Industry appears to be moving that direction.
- Mathworks Extensive support network.
- Mathworks tools for Requirements management, Documentation, and V&V.
- Bus concept makes model management easier.
- Monolithic SystemBuild models not conducive to Reuse and V&V.